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#### ON TRANSVERSE IMPACT ON THE FLEXIBLE FILAMENT

#### **Abstract**

In the paper the construction of solving the problem on transverse impact with constant velocity by obtuse rigid wedge on elastic filament is given. It's assumed that the velocity of wave of strong break is more than the velocity of elastic wave in the filament.

The stress state of an elastic filament at impact on it by solid in some cases essentially depends on geometry of bombarding body and has various applications in many fields of techniques. Beginning from [2] some problems on impact by wedge on flexible filament [3, 4, 6] are solved. In the present paper the solution of the problem on transverse impact by symmetric wedge having plane fore-part (fig. 1) on flexible elastic filament is given.

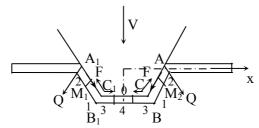


Fig. 1.

§1. Let the transverse impact by rigid symmetric wedge with plane fore-part with the constant velocity V be performed on infinite long flexible linear-elastic rectilinear non-strained filament. It's assumed that after impact a part of the filament  $A_1B_1BA$  covers the surface of the bombarding named wedge, and the velocity of the points A and  $A_1$  are more than the velocity of elastic wave of filament. In addition from the right of the point A and from the left of the point  $A_1$  the filament is at rest state  $\begin{bmatrix} 1,2 \end{bmatrix}$  since  $b = Vctg\gamma > a_0$ . Here  $\gamma$  is an angle between the initial position of filament and check of wedge  $BA(B_1A_1)$ ;  $a_0 = \sqrt{E\rho^{-1}}$  is velocity of elastic wave; b is velocity of the break point  $A(A_1)$ . In the domain  $A_1B_1BA$  in filament four elastic waves whose fronts are the points  $M_1$ ,  $C_1$ , C,  $M_2$ , and two waves of strong break A,  $A_1$   $\begin{bmatrix} 1,2 \end{bmatrix}$  (fig. 1). Denote the length  $BB_1$  by 2L. The behavior of the filament in the domains  $A_1M_1B_1C_1O$  and  $AM_2BCO$  are the same. The velocity of particles of filament in the domains  $OB(OB_1)$  and  $BA(B_1A_1)$  are directed along the check of wedge respectively. In the domains CO and  $C_1O$  the filament is at rest to the time  $t = \frac{L}{a_0}$  relative to wedge.

Since the impact on flexible linear-elastic filament is performed with constant velocity, then in originating domains the filaments determining the parameters are constant. In fig. 1 B,  $B_1$  are stationary break joints and the motion of filaments relative to these breaks are taken as motion via fixed block [5] and the deformations aren't discontinuous at these points. It's assumed that the friction is absent in covering domain

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between the filament and wedge. We'll supply the unknown parameters of motion of flexible filament originating in domains 1, 2, 3, 4, 5, 11, 31 with corresponding indices.

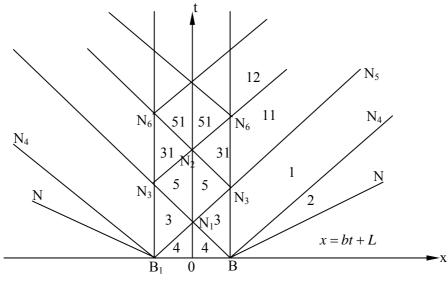


Fig. 2.

The wave picture of motion of filament after impact in the plane (x,t) is shown in fig. 2. The following designations are accepted:  $\varepsilon$  is deformation;  $\vartheta$  is velocity of particles of filament;  $\sigma$  is stress; E is Young's modulus;  $\rho$  is density; t is time; x is Lagrangian coordinate.

At supersonic regime the conditions at the break point A have the form [2]

$$\theta_2 = 0 \; ; \; \varepsilon_2 = \sec \gamma - 1 \; ; \; (\sigma_2 = E\varepsilon_2)$$
 (1.1)

for the first regime when  $F = \mu_* Q$ ;

$$\mathcal{G}_{2} = b(\sec \gamma - 1 - \varepsilon_{2}); \quad b = V \operatorname{ctg} \gamma,$$

$$\varepsilon_{2} = \frac{M^{2}}{M^{2} - tg^{2}\gamma} \left( tg\gamma_{*} - tg\frac{\gamma}{2} \right) \sin \gamma; \quad M = \frac{V}{a_{0}};$$

$$\sigma_{2} = E\varepsilon_{2}; \quad \mu_{*} = tg\gamma_{*},$$
(1.2)

for the second regime when  $F = \mu_* Q$ ,  $\gamma < 2\gamma_*$ ;

$$\theta_{2} = V \cos \gamma (tg \gamma - tg \gamma_{*});$$

$$\varepsilon_{2} = \left( tg \gamma_{*} - tg \frac{\gamma}{2} \right) \sin \gamma;$$

$$\sigma_{2} = 0,$$
(1.3)

for the third regime when  $F = \mu_* Q$ ,  $\gamma > 2\gamma_*$ .

Here F, Q are tangent and normal to the surface of body component of the point force,  $\mu_* = tg \gamma_*$  is a coefficient of coulomb friction at breakpoint.

From (1.3) it follows that  $(\mathcal{G}_2 > 0, \ \mathcal{E}_2 < 0, \sigma_2 = 0)$  for this regime the wrinkling of filament occurs. At first we investigate the behavior of filament when on wave of strong break there are the first and second regimes of motion.

**§2.** On the fronts C,  $M_2$  (fig. 1) or on the fronts 4-3  $(BN_1)$ , 1-2  $(BN_4)$  we have correspondingly

$$\mathcal{G}_3 - \mathcal{G}_4 = -a_0 (\varepsilon_4 - \varepsilon_3), \tag{2.1}$$

$$\mathcal{G}_1 - \mathcal{G}_2 = a_0 (\varepsilon_2 - \varepsilon_1). \tag{2.2}$$

At the point B or on  $BN_3$  (fig. 2) the kinematic condition

$$\mathcal{G}_3 = \mathcal{G}_1 \cos \gamma \tag{2.3}$$

and the condition

$$\varepsilon_1 = \varepsilon_3 \tag{2.4}$$

exist.

Since

$$\theta_4 = 0 ; \quad \varepsilon_4 = 0 ; \quad \theta_3 = a_0 \varepsilon_3 , \quad \left( t < \frac{L}{a_0} \right),$$
(2.5)

then from (2.1)-(2.5) we determine  $\mathcal{G}_1$ ,  $\mathcal{G}_3$ ,  $\mathcal{E}_1$ ,  $\mathcal{E}_3$  in the form of

$$\mathcal{G}_{1} = (1 + \cos \gamma)^{-1} (\mathcal{G}_{2} + a_{0} \varepsilon_{2}); \quad \mathcal{G}_{3} = \mathcal{G}_{1} \cos \gamma; 
\varepsilon_{1} = \varepsilon_{3} = (1 + \sec \gamma)^{-1} (\varepsilon_{2} + \mathcal{G}_{2} a_{0}^{-1}); \quad \sigma_{1} = \sigma_{3} = E \varepsilon_{1}.$$
(2.6)

Note that in the case of the transverse impact by obtuse wedge (2L=0) the parameters of motion of filament in domain 1  $(0 \le x < a_0 t)$  have the form [2]

$$\theta_1^0 = 0; \quad \varepsilon_1^0 = \varepsilon_2 + \theta_2 a_0^{-1}; \quad \sigma_1^0 = E \varepsilon_1^0,$$
 (2.7)

where  $\mathcal{G}_1^0$ ,  $\varepsilon_1^0$ ,  $\sigma_1^0$  mean parameters of filament at impact by obtuse wedge.

Here  $\varepsilon_2$ ,  $\vartheta_2$  are expressed by the formulas (1.1) or (1.2).

It follows from (2.6), (2.7) that the deformation of filament on fronts 1-2 is appreciable less than at transverse impact by obtuse wedge.

At the time  $t = \frac{L}{a_0}$  the elastic waves C and  $C_1$  (fig. 1) meet in point O and two reflective elastic waves 3-5 (fig.2) arise.

It's obvious that if two identical waves move to meet each other, then at the meeting moment (at the point O (fig. 1, fig. 2)) the velocity of particles is zero, and here the stress (deformation) is doubled. This section of filament (the point O) is stationary relative to wedge and we'll consider it as embedded end of filament. Thus in domains 5 (fig. 2)

$$\mathcal{G}_{5} = 0; \, \varepsilon_{5} = 2\varepsilon_{3} = 2(1 + \sec \gamma)^{-1} \left(\varepsilon_{2} + \mathcal{G}_{2} a_{0}^{-1}\right); 
\sigma_{5} = Ee_{5} = E2\varepsilon_{3}.$$
(2.8)

The solutions (2.6), (2.8), (1.1), (1.2) hold when  $\frac{L}{a_0} \le t < \frac{2L}{a_0}$ . At the time  $t = \frac{2L}{a_0}$  after

impact the reflected elastic waves 3-5 (fig. 2) interact with stationary breaks B and  $B_1$ . Since the behaviour of filament with respect to the point O (the axis 0t) (fig. 2) is the same we'll consider a problem at the right hand of filament from the point O. When

 $t = \frac{2L}{a_0}$  from the left of the point B the reflective elastic wave is propagated with front 5-

31  $(N_3N_2)$ , and from right of the point B by BA (fig. 1) the elastic wave is propagated

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with front 11-1  $(N_3N_5)$ . Thus at the period  $\frac{2L}{a_0} \le t < \frac{3L}{a_0}$  after impact on the covering

domain *OBA* (fig. 1) on the filament five domains (domain 5, domain 31, domain 11, domain 1 and domain 2) exist.

The unknown parameters  $\varepsilon$ ,  $\vartheta$  of filament in domains 31 and 11 are determined from the relations

$$\begin{aligned}
\mathcal{S}_5 - \mathcal{S}_{31} &= -a_0 \left( \varepsilon_{31} - \varepsilon_5 \right); \\
\mathcal{S}_{11} - \mathcal{S}_1 &= a_0 \left( \varepsilon_1 - \varepsilon_{11} \right).
\end{aligned} \tag{2.9}$$

The kinematic relation at the point B will be

$$\mathcal{G}_{31} = \mathcal{G}_{11} \cos \gamma \ . \tag{2.10}$$

Condition (2.4) in this case will be of the form:

$$\varepsilon_{11} = \varepsilon_{31} \,. \tag{2.11}$$

Allowing for (2.6), (2.8), from (2.9) we determine the parameters  $\varepsilon_{11}$ ,  $\vartheta_{11}$ ,  $\vartheta_{31}$ ,  $\varepsilon_{31}$  in the form of

$$\varepsilon_{11} = \varepsilon_{31} = \frac{3 + \cos \gamma}{1 + \cos \gamma} \varepsilon_1; \ \sigma_{11} = \sigma_{31} = E \varepsilon_{31}; 
\theta_{11} = a_0 \varepsilon_1 t g^{2} \frac{\gamma}{2} \sec \gamma; \ \theta_{31} = \theta_{11} \cos \gamma.$$
(2.12)

Here  $\varepsilon_1$  is expressed by the formula (2.6).

When  $t = \frac{3L}{a_0}$  fronts of elastic waves 5-31  $(N_3N_2)$  (fig. 2) meet at the point O and are reflected from this point O with fronts 31-51  $(N_2N_6)$ , consequently for the period  $\frac{3L}{a_0} \le t < \frac{4L}{a_0}$  new domain 51 (fig. 2) arise. Since point O is always stationary with respect to bombarding body, then the velocity of particles in domain 51 will be equal to zero.

From the condition on fronts of elastic waves 51-31  $(N_2 - N_6)$ 

$$\theta_{51} - \theta_{31} = a_0 (\varepsilon_{31} - \theta_{51}),$$
 (2.13)

subject to  $\mathcal{G}_{51} = 0$  we determine the deformation  $\varepsilon_{51}$  in the form of

$$\varepsilon_{51} = \varepsilon_{31} + \theta_{31} a_0^{-1} \,. \tag{2.14}$$

Allowing for (2.12) in (2.14) we obtain

$$\varepsilon_{51} = \frac{4\varepsilon_1}{1 + \cos\gamma} \; ; \quad \sigma_{51} = E\varepsilon_{51} \; ; \quad \vartheta_{51} = 0 \; . \tag{2.15}$$

From (2.8), (2.12), (2.15) it follows that the following relation holds

$$\sigma_5 < \sigma_{31} < \sigma_{51} \,. \tag{2.16}$$

Further we can construct a solution of the problem taking into account the multiple reflections of elastic waves from the points O and B.

§3. Now we investigate a problem when on wave of strong break (at the breakpoint A) (fig. 3) we have the condition (1.3), i.e. when at the point A the wrinkling f filament with respect to the point O is symmetric, then we can consider stress state at right hand of filament from the point O. In fig. 4 the wave scheme of motion of filament

in the plane (x,t) is represented. As result of impact at the period  $0 < t < \frac{L}{a}$  in filament (at the right hand) four domains 1, 2, 3, 4 (fig. 3, fig. 4) arise. In domain 4 the filament is

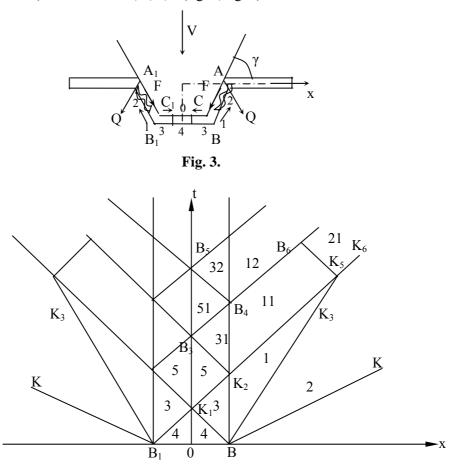


Fig. 4.

at rest. Here the line BK is front of wave of strong break (trajectory of the break point A),  $BK_3$  is a boundary of division of domain with wrinkling (domain 2) and of domain where filament is again stretched (domain 1),  $BK_1$  is front of elastic wave. The straightening front  $BK_3$  is subject to determination in the course of solving the problem, it is propagated with the unknown constant velocity  $\omega$  ( $\omega < a_0$ ). In domain 2 the solution of problem is expressed by the formula (1.3). on the expanding front  $BK_3$  1-2 the following conditions are satisfied

$$\sigma_1 - \sigma_2 = \rho \omega (\vartheta_2 - \vartheta_1); \quad (\sigma_2 = 0); \tag{3.1}$$

$$\mathcal{G}_1 - \mathcal{G}_2 = \omega(\varepsilon_1 - \varepsilon_2). \tag{3.2}$$

Just as we have the conditions (2.3), (2.4) at the point B and the conditions (2.1), (2.5). from the system (2.1), (2.3), (2.4), (2.5), (3.1), (3.2) we determine the unknown parameters  $\omega$ ,  $\varepsilon_1$ ,  $\varepsilon_3$ ,  $\vartheta_3$ ,  $\vartheta_1$  in the form of

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$$\omega = a_1 + \sqrt{a_1^2 - a_2} \; ; \quad \varepsilon_1 = \varepsilon_3 = -\varepsilon_2 \frac{\omega^2}{a_0^2 - \omega^2} \; ;$$
  

$$\vartheta_1 = \vartheta_2 + \frac{a_0^2 \omega}{a_0^2 - \omega^2} \varepsilon_2 \; ; \quad \vartheta_3 = \vartheta_1 \cos \gamma \; ,$$
(3.3)

where

$$a_1 = -\frac{a_0^2 \varepsilon_2}{2(a_0 \varepsilon_2 \sec \gamma - \theta_2)}; \qquad a_2 = \frac{\theta_2 a_0^2}{a_0 \varepsilon_2 \sec \gamma - \theta_2}. \tag{3.4}$$

Note that if impact is performed by an obtuse wedge then straightening velocity will be in the form of [2]

$$\omega_0 = \overline{a}_1 + \sqrt{\overline{a}_1^2 - \overline{a}_2} \; ; \; \overline{a}_1 = \frac{\varepsilon_2 a_0^2}{2\theta_2} \; ; \; \overline{a}_2 = a_0^2 \; .$$
 (3.5)

From (3.4), (3.5) it follows that

$$a_1 < \overline{a}_1 \; ; \quad a_2 < \overline{a}_2 \tag{3.6}$$

and consequently the velocity  $\omega$  is less than the velocity  $\omega_0$  ( $\omega < \omega_0$ ). Here  $\varepsilon_2$  ( $\varepsilon_2 < 0$ ) are determined by the formulas (1.3).

When  $t = \frac{L}{a_0}$  the elastic waves  $B_1K_1$  and  $BK_1$  are reflected from the stationary

point O and new-domain 5 arises ( $K_1K_2$  is reflected elastic wave) (fig. 4) and the solution is domain 5 will be

$$\mathcal{G}_{5} = 0; \quad \varepsilon_{5} = 2\varepsilon_{3} = -\frac{2\varepsilon_{2}\omega^{2}}{a_{0}^{2} - \omega^{2}}; \quad (\varepsilon_{2} < 0);$$

$$\sigma_{5} = 2\rho a_{0}^{2} \varepsilon_{3}.$$
(3.7)

At the period  $t = \frac{2L}{a_0}$  the fronts of reflected elastic wave  $K_1K_2$  interact with the stationary break B and from the left of the point B the reflective elastic wave with the front  $K_2B_3$  is propagated, and from the right the elastic wave with the front  $K_2K_4$  (fig. 4) is propagated.

In the domain *OBA* (fig. 3) in the filament for the period  $\frac{2L}{a_0} \le t < \frac{3L}{a_0}$  five domains arise and they are denoted by 2, 1, 11, 31, 5. The solutions in domains 2, 1, 5 are known, we have to determine solutions in domains 11 and 31.

The solutions of problem in domains 11 and 31 are determined from conditions on the fronts  $K_1B_3$ ,  $K_2K_4$  and have the form (2.9), consequently, the parameters in these domains are determined by the formulas (2.12), but here  $\varepsilon_1$ ,  $\vartheta_1$  are expressed by the formulas (3.3), (3.4). When  $t = \frac{3L}{a_0}$  the front of the elastic wave  $K_2B_3$  is reflected from the point O with front  $B_3B_4$  and new domain 51 (fig. 4) arises. The unknown parameters

 $\varepsilon_{51}$ ,  $\vartheta_{51}$  in domain 51 are determined by the formulas (2.15), but only  $\varepsilon_{1}$  is expressed by the relations (3.3), (3.4). since the velocity of elastic wave  $a_{0}$  is more than the velocity of straightening wave  $\omega$ , then after some period the front of elastic wave  $K_{2}K_{4}$  overtakes

the straightening wave  $BK_3$  and at the point  $K_5$  interaction of these waves occurs. The interaction time is determined from the relation

$$\omega t + L = a_0 (t - t_0); \ t_0 = \frac{2L}{a_0}$$
 (3.8)

and

$$t = t_1 = \frac{3L}{a_0 - \omega} \,. \tag{3.9}$$

Note that for the given L, the period  $t=t_1$  and consequently wave picture of motion depends on the relation  $\frac{\omega}{a_0}$ . Let  $t_1>\frac{4L}{a_0}$ , i.e.  $\frac{1}{4}<\frac{\omega}{a_0}<1$ , then for  $t=\frac{4L}{a_0}$  the front of elastic wave  $B_3B_4$  interacts with stationary break  $B_3$  and from the left of the point  $B_3$  the elastic wave in the form of  $B_4B_5$  is reflected, from the right the elastic wave with the front  $B_4B_6$  (fig, 4) is propagated. After the elastic wave  $B_3B_4$  is reflected from the point  $B\left(t>\frac{4L}{a_0}\right)$ , the front of elastic wave  $K_2K_4$  meet at the point  $K_5$  with the front of expanding wave  $BK_3$ . The front of elastic wave  $K_2K_4$  is reflected in the form of  $K_5B_6$ , at that time the elastic wave  $K_2K_4$  passing through front  $BK_5$  and propagating in

of expanding wave  $BK_3$ . The front of elastic wave  $K_2K_4$  is reflected in the form of  $K_5B_6$ , at that time the elastic wave  $K_2K_4$  passing through front  $BK_5$  and propagating in wrinkling domain doesn't influence to behaviour of filament at the break point A, however it influences to the expanding front  $K_5K_6$  (fig. 4), whose velocity is again remains unknown, and we must determine it in the course of solving the problem.

Thus new domains 32, 12, 21 (fig. 4) arise.

For determination of the unknown parameters  $\varepsilon_{32}$ ,  $\vartheta_{32}$ ,  $\varepsilon_{12}$ ,  $\vartheta_{12}$  at the point B and on the fronts  $B_4B_5$ ,  $B_4B_6$  we have the following relations

$$\begin{aligned} \mathcal{G}_{32} &= \mathcal{G}_{12} \cos \gamma \; ; \; \varepsilon_{32} = \varepsilon_{12} \; ; \\ \mathcal{G}_{51} &- \mathcal{G}_{32} = -a_0 \left( \varepsilon_{32} - \varepsilon_{51} \right) ; \\ \mathcal{G}_{12} &- \mathcal{G}_{11} = a_0 \left( \varepsilon_{11} - \varepsilon_{12} \right). \end{aligned}$$
(3.10)

Form (3.10) with regard  $\theta_{51} = 0$  we obtain

$$\varepsilon_{32} = \varepsilon_{12} = \frac{1}{1 + \cos \gamma} \left[ \left( \frac{\vartheta_{11}}{a_0} + \varepsilon_{11} \right) \cos \gamma + \varepsilon_{51} \right];$$

$$\vartheta_{12} = \frac{1}{1 + \cos \gamma} \left( \vartheta_{11} + a_0 \varepsilon_{11} - a_0 \varepsilon_{51} \right); \quad \vartheta_{32} = \vartheta_{12} \cos \gamma.$$
(3.11)

Here  $\mathcal{G}_{11}$ ,  $\varepsilon_{11}$ ,  $\varepsilon_{51}$  are expressed by the formulas (2.12), (2.18), but  $\varepsilon_{1}$  participating in these formulas is determined from (3.3), (3.4). Now we determine  $\mathcal{G}_{21}$ ,  $\varepsilon_{21}$  in domain 21 and the velocity of expanding wave  $\omega_{*}$ . For this we have the common conditions on the front  $K_{5}K_{6}$  in the form of

$$\sigma_{21} - \sigma_2 = \rho \,\omega_* (\mathcal{G}_2 - \mathcal{G}_{21});$$
  

$$\mathcal{G}_{21} - \mathcal{G}_2 = \omega_* (\mathcal{E}_2 - \mathcal{E}_{21})$$
(3.12)

and the continuity condition of displacement on the front in the form of

$$\mathcal{G}_{11} - \mathcal{G}_{21} = -a_0 (\varepsilon_{21} - \varepsilon_{11}).$$
(3.13)

From (3.12), (3.13) we determine  $\omega_*$ ,  $\varepsilon_{21}$ ,  $\vartheta_{21}$  in the form of

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$$\omega_{*} = a_{0} \left( \sqrt{\frac{\varepsilon_{2}^{2} a_{0}^{2}}{4} + b_{1} b_{2}} + \frac{\varepsilon_{2} a_{0}}{2} \right) b_{1}^{-1};$$

$$\varepsilon_{21} = \frac{1}{\omega_{*} + a_{0}} \left( \theta_{2} + \omega_{*} \varepsilon_{2} + a_{0} \varepsilon_{11} - \theta_{11} \right);$$

$$\theta_{21} = \frac{1}{\omega_{*} + a_{0}} \left[ \left( \theta_{11} - a \varepsilon_{11} \right) \omega_{*} + \theta_{2} a_{0} + \omega_{*} a_{0} \varepsilon_{2} \right],$$
(3.14)

where

$$b_1 = \theta_2 - \theta_{11} + a_0(\varepsilon_{11} - \varepsilon_2); \ b_2 = \theta_2 - \theta_{11} + a_0\varepsilon_{11}.$$
 (3.15)

Let  $\mu_* = 0.2126$ ;  $\gamma = 36^\circ$ ;  $a_0^{-1} ctg \gamma = 1.2$ ; M = 0.8727, then from (3.3) and (3.14) we correspondingly obtain

$$\omega a_0^{-1} = 0.887$$
;  $\omega_* a_0^{-1} = 0.9098$ . (3.16)

From (3.16) it follows that interaction of elastic wave with front of the straightening wave influences to increase of velocity of straightening wave. The solution of problem with regard to multiple reflections is not difficult.

Note that for the determined combinations of parameters of problem at transverse impact by obtuse wedge the filament is broken at the point of impact x = 0 at the moment t = 0 under the limiting condition [2]

$$\sigma = \sigma_{np} \; ; \; \left( \sigma = 2\rho \, a_0^2 \varepsilon_1^0 \right), \tag{3.17}$$

where  $\sigma_{np}$  is ultimate strength (failure stress) of filament.

But for the same original parameters of the problem at impact by wedge with plane fore-part when t=0 in filament the failure stress doesn't arise. Consequently, the filament may be broken after some period and this case in the present paper isn't considered.

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